

# Measurement and Identification of the Nonlinear Dynamics of a Jointed Structure Using Full-Field Data; Part II - Nonlinear System Identification

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## Abstract

The dynamic responses of assembled structures are greatly affected by the mechanical joints, which are often the cause of nonlinear behavior. To better understand and, in the future, tailor the nonlinearities, accurate methods are needed to characterize the dynamic properties of jointed structures. In this paper, the nonlinear characteristics of a jointed beam is studied with the help of multiple identification methods, including the Hilbert Transform method, Peak Finding and Fitting method, Dynamic Mode Decomposition method, State-Space Spectral Submanifold, and Wavelet-Bounded Empirical Mode Decomposition method. The nonlinearities are identified by the responses that are measured via accelerometers in a series of experiments that consist of hammer testing, shaker ringdown testing, and response/force-control stepped sine testing. In addition to accelerometers, two high-speed cameras are used to capture the motion of the whole structure during the shaker ringdown testing. Digital Image Correlation (DIC) is then adopted to obtain the displacement responses and used to determine the mode shapes of the jointed beam. The accuracy of the DIC data is validated by the comparison between the identification results of acceleration and displacement signals. As enabled by full-field data, the energy-dependent characteristics

35 of the structure are also presented. The setup of the different experiments is described in  
36 detail in Part I of this research. The focus of this paper is to compare nonlinear system  
37 identification methods applied to different measurement techniques and to exploit the use  
38 of high spatial resolution data.

39 *Keywords:* Jointed Structures, Nonlinear System Identification, Digital Image Correlation  
40 (DIC), Amplitude-Dependent Characteristics, Energy-Dependent Characteristics

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## 41 1. Introduction

42 Mechanical joints are of crucial importance in the dynamics of assembled structures  
43 [1]. Significant nonlinearities can be caused by friction, wear, and non-idealized boundary  
44 conditions [2, 3, 4]. With slip and clapping behaviors happening between the interfaces  
45 [5], the complexity of the joints makes the modeling highly challenging. Therefore, it is  
46 extremely important to characterize the behaviors of jointed structures experimentally with  
47 high accuracy, which is a pressing need in several industries such as aerospace [6], automotive  
48 [7], and naval [8].

49 Among several tools available, backbone curves are a common method for describing  
50 nonlinear systems [9, 10] as they characterize the amplitude-dependent natural frequency  
51 and damping for single mode oscillations from free decay data [11]. The interactions that  
52 may occur in the system can be elucidated, enabling the investigation of modal energy  
53 exchange due to the nonlinearities, which cannot be analyzed by conventional linear system  
54 identification methods. Moreover, the backbone curve makes it possible to estimate or  
55 update the nonlinear characteristics within a model based on the experimental response [12,  
56 13, 14, 15]. Even more so than frequency, damping is critical in predicting and understanding  
57 the behaviors of a structure [16, 17]. Often, the damping nonlinearity in a jointed structure  
58 changes the damping capacity of a structure by orders of magnitude across a range of  
59 response amplitudes while the stiffness nonlinearity changes the natural frequencies by a few  
60 percent [18]. The amplitude-damping curve is a general way to demonstrate the dissipation  
61 of systems [10], especially for jointed structures [1]. Together, the knowledge of amplitude-  
62 dependent damping and amplitude-dependent frequency characteristics can be used to make  
63 predictions for the steady-state forced harmonic response of a structure. The key to this  
64 is the single-nonlinear-mode theory, for which simple, explicit approximations are available  
65 [19, 20, 21].

66 There are many different methods for calculating the amplitude-dependent frequency and  
67 damping curves, hereafter also referred to as the frequency and damping backbone curves.  
68 The Hilbert Transform (HT) method [22, 23] is generally considered to be a reliable time-  
69 domain identification method [9] and relatively accurate compared with other identification  
70 methods [24]. The Peak Finding and Fitting (PFF) method [25] is another time-domain  
71 method, which is recently proposed based on the traditional zero-crossing (ZC) method  
72 [10]. One limitation of the above two methods is that they are mono-modal, i.e., they  
73 are formulated to analyze one mode and one point at a time and cannot identify modal

74 interactions. One approach to address this shortcoming is the use of frequency-domain  
75 methods, such as the Dynamic Mode Decomposition (DMD) method, which finds coherent  
76 spatial-temporal pattern from the dataset [26]. When the dimension of acquisition is less  
77 than the rank of the system, the modified version of DMD, i.e., the Hankel DMD [27, 28],  
78 can be used to analyze the system. For analyzing vibrations of mildly nonlinear systems, the  
79 window based Hankel DMD can be used to obtain the time-varying modal parameters. A  
80 fundamentally different approach from time-domain and frequency-domain methods is the  
81 identification of the nonlinear normal modes of a system. Nonlinear normal modes (NNMs)  
82 [29] are a popular and useful tool for the analysis of vibrations in nonlinear, conservative  
83 systems. Similarly, Spectral Submanifolds (SSMs) [30] represent the extension of classical  
84 nonlinear normal modes to forced-damped nonlinear structures. SSMs can be exploited to  
85 gain precious insights on the dynamical behavior [31, 32] and to construct reduced-order  
86 models [33]. A methodology was proposed to analytically compute both the shape of SSMs  
87 and their corresponding backbone curves according to a data-assimilating model that was  
88 fitted on experimental vibration signals [34]. This means that a data-driven polynomial  
89 state space model can be constructed based on SSMs. Besides the aforementioned methods,  
90 time-frequency analysis is another powerful way to extract information about frequency and  
91 damping in structural dynamics. Empirical Mode Decomposition (EMD) is a technique  
92 that decomposes a signal into a finite set of intrinsic mode functions (IMFs) [35]. Wavelet-  
93 Bounded EMD (WBEMD) is an improvement over EMD, improving the separation of IMFs,  
94 which represent different characteristic time scales. In this manner, the WBEMD method  
95 is able to identify not just frequency and damping information, but also the portions of the  
96 time record that the information is extracted from, allowing an analyst to visualize how the  
97 system changes over time across a large range of frequencies [36]. These system identification  
98 methods are described more in detail in Sect. 2.

99 Nonlinearities are usually characterized by amplitude-dependent curves [9, 37], i.e., the  
100 frequency/damping is plotted as the function of the amplitude. The shortcoming of this  
101 representation is that the characterization of the nonlinearity is related to the amplitude  
102 of a certain accelerometer, thus a specific point/location. In the view of reaching a more  
103 global metric, attention has been paid to the relationship between the energy stored in a  
104 mode and the corresponding frequency characteristics (hereafter referred to as the frequency-  
105 energy relationship) [38]. While frequency-energy plots can be easily computed in analytical  
106 or numerical studies [29, 39], their experimental determination remains a challenge. The  
107 energy of a physical structure is difficult to calculate based on the information of either  
108 dense or sparse accelerometers.

109 With the development of high-speed cameras, the motion of the whole structure can be  
110 recorded by images with high spatial resolution. Digital Image Correlation (DIC) [40, 41]  
111 can then process the recorded images and extract displacements of the structure. Although  
112 they have lower accuracy, high-speed cameras perform non-contact measurements and al-  
113 low higher spatial resolution than accelerometers. DIC has recently been applied to study  
114 jointed structures [5, 42] by measuring the slip and separation behaviors in the interfaces.  
115 Additionally, DIC has advantages over other non-contact measurement techniques, chiefly

116 laser Doppler vibrometer (LDV). For instance, DIC is often a more affordable approach  
117 than LDV and the latter is also more time consuming, since a LDV has to scan across the  
118 surface one point at a time. The disadvantage of current DIC methods, though, is the large  
119 post-processing time to extract the large amount of data from videos of the experiment as  
120 well as limitations in memory for data storage.

121 The number of testing methods for extracting amplitude-dependent characteristics from  
122 nonlinear structures has been growing in the last two decades [38]. Hammer excitation  
123 or imposed initial conditions are used to trigger decaying oscillations [24], but they typ-  
124 ically lack in isolating single mode responses. Another family of methods, such as force  
125 appropriation [43, 44] or control-based continuation [45], exploits mono-harmonic forcing  
126 from a shaker as well as the phase-quadrature criterion to isolate resonant vibrations and  
127 to extract amplitude-dependent properties. However, these methods can be excessively  
128 time-consuming and they can suffer from structure-shaker interactions. Resonance decay,  
129 or shaker ringdown testing [46, 47], has been introduced to solve these issues. Here, a  
130 high-amplitude resonant vibration is isolated via phase-quadrature using the shaker and,  
131 afterwards, the shaker is decoupled from the structure, which shows single-mode decaying  
132 responses. Although decoupling the shaker from the structure is not a trivial operation, the  
133 extraction of amplitude-dependent properties is faster with respect to other shaker-based  
134 methods.

135 This research consists of two parts. The first [48] focuses on experimental investigations,  
136 describing the tools, methods, and challenges for extracting the data from a benchmark  
137 system in vibrations of joint-assembled structures. Both accelerometers and high-speed  
138 cameras are adopted to record the motion of the system over time. In this work, the spatially  
139 dense data from [48] is analyzed in order to assess the efficacy of using DIC for nonlinear  
140 system identification. Multiple nonlinear system identification methods are utilized and  
141 compared with each other in order to establish their performances and to highlight pros and  
142 cons from a practitioner viewpoint. Both free decay vibration and forced-response testing  
143 are analyzed for the extraction of backbone and damping curves and, due to the availability  
144 of full-field data and energy estimations, their interconnections are investigated with novel  
145 metrics. Moreover, this research shows the agreement of data from the accelerometers and  
146 DIC and sheds further light on amplitude-dependent properties by inspecting the shape  
147 variations and the energy of the structure.

148 The paper is organized as follows. In Sect. 2, the different methods adopted are summa-  
149 rized. Section 3 reports on the analysis of experiments, first focusing on accelerometer data  
150 and then on DIC data, which comes from hammer testing, shaker ringdown testing, and  
151 response/force-control stepped sine testing. Additionally, an energy criterion is established  
152 to connect amplitude-dependent properties from forced-response testing with those obtained  
153 via free-decay response. Section 4 presents a critical comparison of the adopted methods,  
154 and Sect. 5 concludes the paper.

## 155 2. Nonlinear System Identification Methods

156 In this section, the five methods utilized throughout this research are briefly reviewed,  
157 including the Hilbert Transform (HT) method [22, 23], Peak Finding and Fitting (PFF)  
158 method [25], State-Space Spectral Submanifold (SS-SSM) method [34], Dynamic Mode De-  
159 composition (DMD) method [27, 28], and Wavelet-Bounded Empirical Mode Decomposition  
160 (WBEMD) method [36]. Among these methods, the HT and PFF are time-domain ap-  
161 proaches, the DMD belongs to frequency-domain method, the WBEMD is a time-frequency  
162 analysis method [9], and the SS-SSM is a data-driven modeling technique. Each of these  
163 methods are applied to analyze the data from different experiments in Sect. 3.

### 164 2.1. The Hilbert Transform Method

165 Generally, the HT method is regarded as one of the most popular and reliable nonlinear  
166 system identification methods. It was proposed by Feldman [22, 23] and was applied to  
167 identify different nonlinear systems [49, 50, 51]. The sum of the signal and its HT composes  
168 the analytical signal. When the analytical signal is written in polar coordinates, the in-  
169 stantaneous amplitude and instantaneous frequency can be calculated. However, smoothing  
170 is needed for the identified characteristics [52]. There are a few ways to accomplish the  
171 smoothing, including filtering the identification results and mirroring the original signal be-  
172 fore applying the HT method. According to [24], Empirical Modal Decomposition (EMD)  
173 [35] is an effective way that can be combined with the HT method to provide the smoothed  
174 results for both instantaneous amplitude and frequency. Based on the local time scale of  
175 the signal, EMD can be used to decompose the signal into intrinsic mode functions [35], the  
176 first of which generally contains the highest frequency components. By excluding the high  
177 frequency components, EMD is able to smooth the results identified by the HT method.  
178 This means that EMD serves as a lowpass filter, but, importantly, it does not introduce a  
179 phase shift [24]. After smoothing, the damping ratio can be calculated based on the two  
180 smoothed quantities.

### 181 2.2. The Peak Finding and Fitting Method

182 The PFF method [25] is based on the zero-crossing (ZC) method [10], which calcu-  
183 lates instantaneous frequency according to the time instants where the signal crosses zero.  
184 The identification accuracy of PFF was demonstrated by several cases in [24]. In the PFF  
185 method, the local maxima (and minima) of the signal are first searched. With the combi-  
186 nation of the local maxima (and minima) and the two points near them, the signal near the  
187 local maxima (and minima) can be approximated by a polynomial. Therefore, improved  
188 estimations of the true values of the peaks (and valleys) are obtained from the fitted polyno-  
189 mial, which can overcome errors due to under-sampling. Then, the PFF method calculates  
190 instantaneous frequency as the reciprocal of the difference between two sequential time in-  
191 stants at which the signal reaches these peaks (and valleys). The PFF method is able to  
192 provide the result of the instantaneous frequency that achieves similar accuracy as the ZC

193 method. Furthermore, by improving the extraction of the peaks, the instantaneous ampli-  
194 tude of the response can be obtained with higher accuracy by the PFF method. Therefore,  
195 the damping ratio, which is calculated by the instantaneous frequency and instantaneous  
196 amplitude, has less noise compared with that obtained by the ZC method [25]. Moreover,  
197 unlike the HT method, the instantaneous frequency and instantaneous amplitude calculated  
198 by the PFF method do not need to be smoothed, thus saving computational time.

### 199 *2.3. The State-Space Spectral Submanifold Method*

200 A spectral submanifold (SSM) is referred to as the smoothest nonlinear continuation of a  
201 linearized mode shape of an equilibrium [30]. Thus, SSMs are calculated in order to analyze  
202 the reduced dynamics of the system, giving a nonlinear extension of the linear dynamics of  
203 the modal subspace. Moreover, the amplitude-dependent frequency curve along SSMs has  
204 been shown to approximate the backbone curve of forced frequency responses [11]. With  
205 respect to methods developed in the framework of conservative nonlinear normal modes  
206 [29], the theory of SSMs relaxes some critical assumptions on the purely parasitic effect of  
207 damping. Thus, the SS-SSM data-driven method developed in [34] puts no restriction on the  
208 type and magnitude of the damping; however, it does require that the signal being analyzed  
209 have a dynamic response that is sufficiently close to a single-mode (or two-dimensional) SSM  
210 of the underlying system. The data are simultaneously assimilated into a nonlinear discrete-  
211 time state space model from which the parametrization and the reduced SSM dynamics are  
212 analytically computed, giving a nonlinear extension of the linear dynamics of the modal  
213 subspace. The reduced-order model of the SSM is then exploited to derive amplitude-  
214 dependent properties. This approach was demonstrated to reproduce backbone curves with  
215 high accuracy according to the results of both numerical and experimental data [34, 53].

### 216 *2.4. The Dynamic Mode Decomposition Method*

217 For the fluid dynamics community, DMD is one of the most promising system identifica-  
218 tion tools since it allows for high dimensional measurements. This algorithm finds coherent  
219 spatial-temporal patterns from a densely measured dataset [26]. In this paper, since dense  
220 spatial data is provided due to the combination of high-speed cameras and DIC, DMD is  
221 used to comprehend the underlying dynamics of the structure. As the rank of the structural  
222 vibration is much less than the dimension of data acquisition, this type of system identi-  
223 fication technique is well suited. Also, for the cases when only one accelerometer is used,  
224 this technique is still able to provide accurate information about the modal parameters of  
225 the system. In the present context, the modified version of DMD, known as Hankel DMD  
226 [27, 28], is used to analyze the system. The formulation of Hankel DMD is similar to the  
227 formulation of the Eigensystem Realization Algorithm [28, 54] that is typically used for  
228 system identification of linear dynamical systems. Hence, in this study, for analyzing the  
229 vibration of the jointed structure that is a mildly nonlinear system, window-based Hankel  
230 DMD is used to capture the time-varying modal parameters of the structure. It is assumed  
231 that in a small time window, this mildly nonlinear structure approximately behaves like a

232 linear dynamical system. Amplitude information is obtained via filtering at frequencies of  
233 interest.

### 234 *2.5. The Wavelet-Bounded Empirical Mode Decomposition Method*

235 By exploiting EMD, a signal is decomposed into a finite set of nearly orthogonal, monochro-  
236 matic intrinsic mode functions (IMFs) [35, 55]. Theoretically, every IMF should be a physical  
237 and mathematical representation of a single characteristic time scale contained in the origi-  
238 nal oscillatory signal. However, in practice, EMD yields spurious, non-physical IMFs, which  
239 need to be eliminated before starting the dynamical analysis [56]. Another issue with the  
240 regular EMD is the problem of mode mixing, where one IMF is comprised of components at  
241 different frequencies and, thus, is not a fair representation of any single time scale extracted  
242 from the original signal. The WBEMD augments the regular EMD by using a masking sig-  
243 nal, which provides a quantitative measure of the isolation of an IMF around a characteristic  
244 frequency [36]. This measure is used as the objective function of a minimization problem.  
245 In the WBEMD algorithm, after applying EMD, the IMF is transformed into the maximum  
246 wavelet domain [57] where a bounding function is fitted over the IMF. This procedure is  
247 able to extract optimally-separated IMFs, each representing distinct time scales present in  
248 the signal. Once frequency information is obtained, ad-hoc filtering is applied to retrieve  
249 amplitude and damping characteristics of the signal.

## 250 **3. Analysis of Experiments**

251 Using the nonlinear system identification methods described in the previous section, the  
252 experiments on the Half Brake-Reuß Beam (HBRB) described in [48] are analyzed. These  
253 experiments include hammer testing, shaker ringdown testing, and response/force-control  
254 stepped sine testing. **Figure 1** summarizes the experiments, along with the pre-processing  
255 techniques and the identification method used to extract nonlinear information. The HBRB  
256 is modified from a normal BRB [1], which is a prismatic 304 stainless steel beam joined  
257 in the middle through a three-bolt lap joint (**Fig. 2**), to be half of the nominal thickness.  
258 This modification was chosen in order to reduce the first three natural frequencies of the  
259 system such that high-speed videography could easily measure the response of the first three  
260 modes without sacrificing spatial resolution. The nonlinear characteristics extracted from  
261 the hammer testing and shaker ringdown testing are compared with the backbone curves  
262 extracted directly from the response/force-control stepped sine testing [48].

263 During the shaker ringdown testing, apart from accelerometers, two high-speed cameras  
264 are used to capture images of the HBRB. DIC [40, 41] is applied to the images, providing  
265 the displacement responses of the beam. The specific set of measurements used from each  
266 experiment are further highlighted in the flowchart of **Fig. 1**. During this study, each bolt is  
267 tightened to 10 N·m and the first mode is mainly considered (except where noted otherwise).  
268 Further details of all of the experiments in this section can be found in Part I of this research  
269 [48].

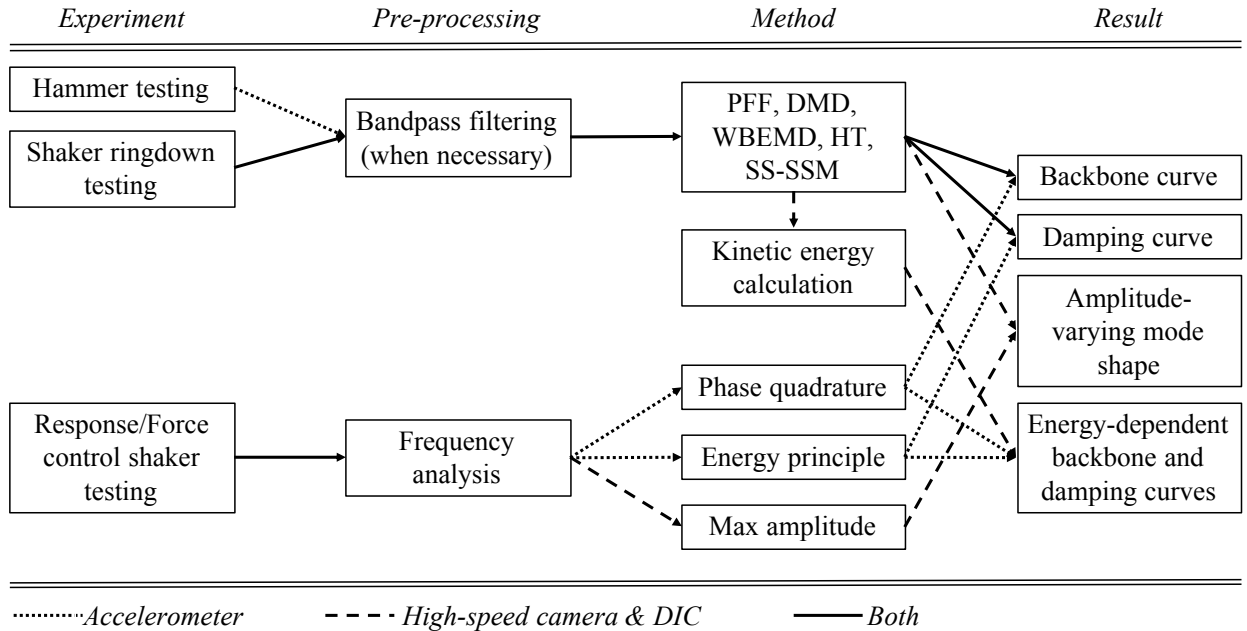


Figure 1: Flowchart that represents the experiments and the methods discussed in this paper, also including the necessary pre-processing and the identification results. Different measurements, from accelerometer or speed camera, are highlighted using different linestyles.

### 270 3.1. Free Decay Response by Accelerometers

271 In this section, the identification results from free decay responses of accelerometer mea-  
 272 surements are discussed. Specifically, the amplitude-dependent properties extracted via  
 273 hammer testing and shaker ringdown using the methods presented in Sect. 2 are discussed.

#### 274 3.1.1. Hammer Testing

275 During the hammer testing of the HBRB, free boundary conditions are achieved by  
 276 hanging the beam with the combination of fishing lines and bungees [48, 15]. The constraints  
 277 (at the node points of the first mode) are applied since the beam should be confined within a  
 278 limited space, and it is demonstrated in Part I that the specific arrangements of constraints  
 279 used here do not significantly affect the characteristics of the beam [48]. The HBRB is  
 280 excited by the hammer and is measured by three accelerometers (**Fig. 2**) with a sampling  
 281 frequency of 6400 Hz. As an example, a free decay response obtained by the accelerometer  
 282 nearest to the impact location (indicated as Accel 3 in **Fig. 2**) is shown in **Fig. 3** (a).

283 With the time history response, the methods described in **Sect. 2** are applied to obtain  
 284 backbone curves and amplitude-damping curves of the HBRB, with the exception of the  
 285 SS-SSM method. This method is excluded since it assumes that the data are dominated by  
 286 a single-mode SSM, which is not the case in hammer testing.

287 The backbone curves extracted in this section are defined as the amplitude-frequency  
 288 relations of principal modes, or components, of the decaying signals. Since the HT and



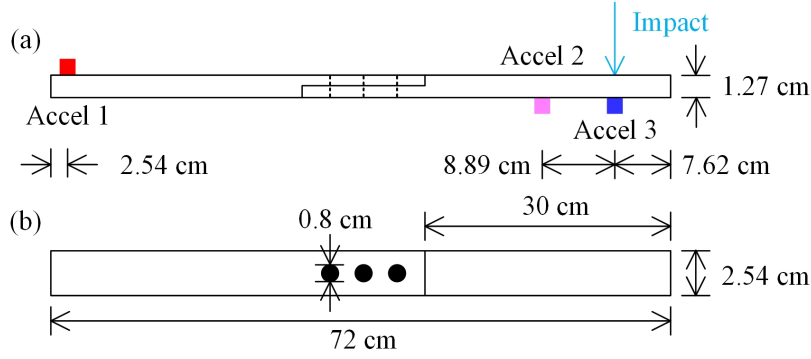


Figure 2: The setup of the hammer testing for the HBRB, showing from the (a) top and (b) side views.

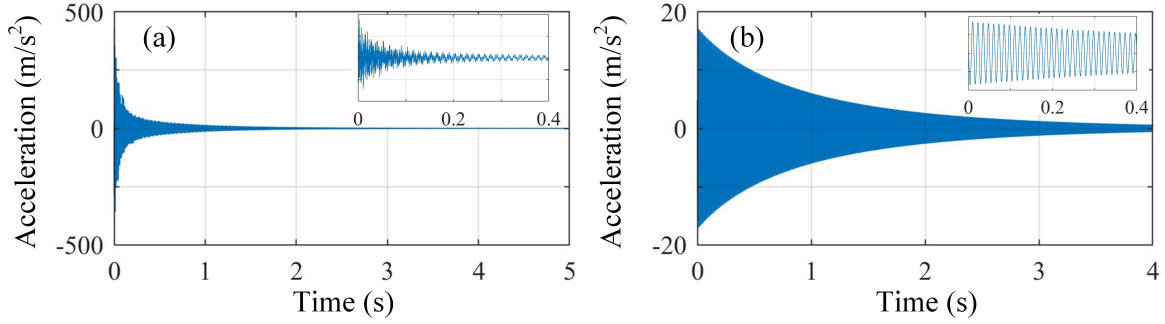


Figure 3: The free decay response measured by Accel 3, showing (a) raw response (with zoomed response from 0 s to 0.4 s) and (b) filtered response (with zoomed response from 0 s to 0.4 s).

289 PFF methods can only be applied to a signal with one frequency component, the free decay  
 290 response has to be filtered [58] before these two methods are applied. The effects of applying  
 291 a third-order butterworth bandpass filter whose frequency band is from 75.27 Hz to 85.27  
 292 Hz are shown in **Fig. 3 (b)**. The acceleration response is truncated at 4 s to obtain the  
 293 identification results with a high signal-to-noise ratio. Additionally, for the HT method, the  
 294 signal is further truncated from 0.5 s to 3.9 s so that the end effects can be reduced.

295 As shown in **Fig. 4**, all of the different methods show great agreement in the identification  
 296 of both the backbone curve and amplitude-damping curve for the first mode. This means  
 297 that the instantaneous frequency, amplitude, and damping ratio are well identified by these  
 298 methods. The change in damping (from 0.13% to 0.25%) as the response amplitude increases  
 299 from 0.4 m/s<sup>2</sup> to 17 m/s<sup>2</sup> is significantly greater than the change in frequency (from 80.3  
 300 Hz to 80 Hz), which is typical in the context of jointed structures.

301 The spikes observed in the damping curve obtained by the HT method are caused by  
 302 the calculation of the damping ratio, which is related not only to the amplitude, but also  
 303 to its derivative [25]. Thus, although the signal was already truncated before using the HT  
 304 method, there are still errors at the two ends in the damping ratio, which is one of the  
 305 drawbacks of the HT method.

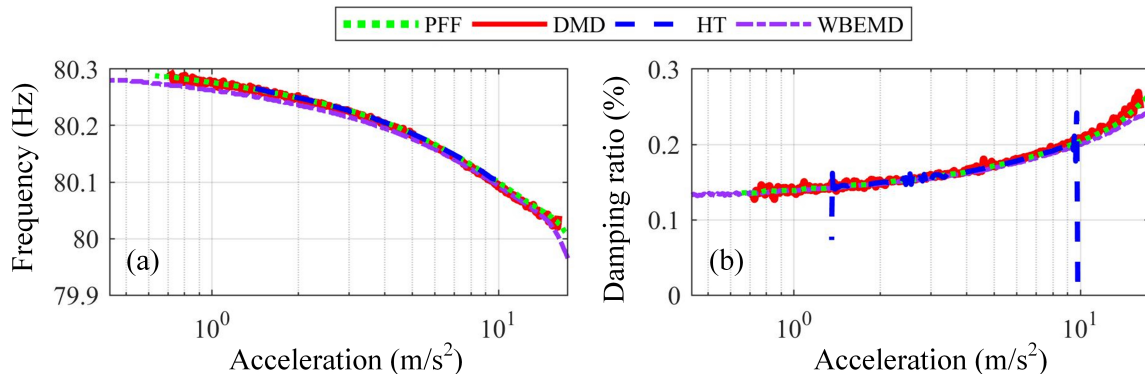


Figure 4: The identification results of the hammer testing by the PFF (green dotted line), HT (blue dashed line), DMD (red solid line), and WBEMD (purple dash-dotted line) methods, showing (a) backbone curve and (b) amplitude-damping curve.

306 In comparing the PFF, DMD, and WBEMD methods, the PFF and DMD method both  
 307 overlay the HT method, while the WBEMD method results in slightly lower predictions,  
 308 both for frequency and damping. This is justified by the fact that WBEMD does not pre-  
 309 filter the signals, in contrast with the other methods. Lastly, both the PFF and WBEMD  
 310 show significantly less noise, especially in measurements of damping, than the DMD and  
 311 HT methods.

### 312 3.1.2. Shaker Ringdown Testing

313 With the experimental setup similar to that of the hammer testing, shaker ringdown  
 314 testing for the first mode of the HBRB is performed. The shaker is attached to the same  
 315 location as the hammer impact point in **Fig. 2** and it is used to obtain approximately single-  
 316 mode responses [59]. After isolating a resonant frequency response (reaching a 90-degree  
 317 phase lag between force and driven point) or NNM, the shaker is physically detached from  
 318 the HBRB [60], and the system decays to the equilibrium along the related SSM [30], as  
 319 illustrated in **Fig. 5**. See [48] for a more detailed description of the experiment. Hence,  
 320 the backbone curve computed is the SSM free decay instantaneous frequency-amplitude  
 321 relation. Apart from the three accelerometers, two high-speed cameras are used to measure  
 322 the displacement of the beam. DIC [40, 41] is applied to the captured images to acquire the  
 323 displacement responses. The analysis of the displacement responses is presented in Sect. 3.3,  
 324 while this section focuses on the acceleration measured by the accelerometer located on the  
 325 shaker excitation point.

326 The Short-time Fourier transform (STFT) analysis is conducted on the response of the  
 327 shaker ringdown testing. As it can be seen from **Fig. 5** (b), the energy is only in the  
 328 first mode (below 100 Hz) and it stays in the first mode, except at the moment when  
 329 the shaker is decoupled from the beam; however, the effect shown by the STFT analysis  
 330 (Fig. 5(b)), does not persist past 30.2 seconds, which is where the signal is truncated for  
 331 analysis. All of the methods described in **Sect. 2** are applied to the signal from time 30.2 s

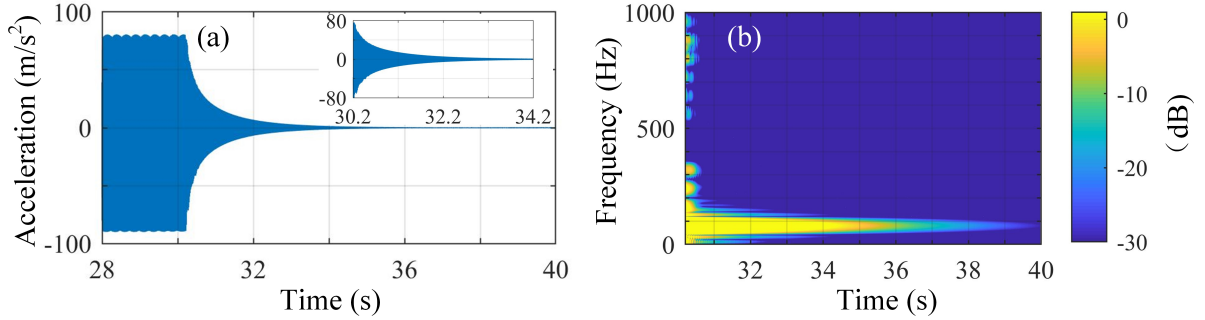


Figure 5: The acceleration response measured by Accel 3, showing (a) raw response from 28 s to 40 s (with zoomed response from 30.2 s to 34.2 s) and (b) STFT analysis of the response from 30.2 s to 34.2 s.

332 to 40 s (i.e., from the moment that the shaker is decoupled from the system until the noise  
 333 floor is reached). It is shown in **Fig. 6** that there is little difference in the identification  
 334 results, which means that the instantaneous frequency, amplitude, and damping ratio are  
 335 all identified with similar accuracy by these methods, with several exceptions discussed in  
 336 what follows. The instantaneous damping ratio rises with the amplitude increasing, from  
 337 about 0.1% to more than 0.6% (**Fig. 6** (b)). Thus, a strong nonlinearity in the damping  
 338 of the HBRB is observed. The instantaneous frequency undergoes small variations (0.5 Hz,  
 339 or 0.6%) in this amplitude regime. Unlike the impact hammer tests, which were limited  
 340 in terms of the amount of energy that they could put into a single mode and could not  
 341 drive the system to a high enough amplitude to make this observation, the shaker ringdown  
 342 experiments exhibit a softening-hardening trend as the amplitude increases (**Fig. 6** (a)). As  
 343 shown in the next sections, this trend is not a characteristic of this specific accelerometer,  
 344 but rather of the whole structure.

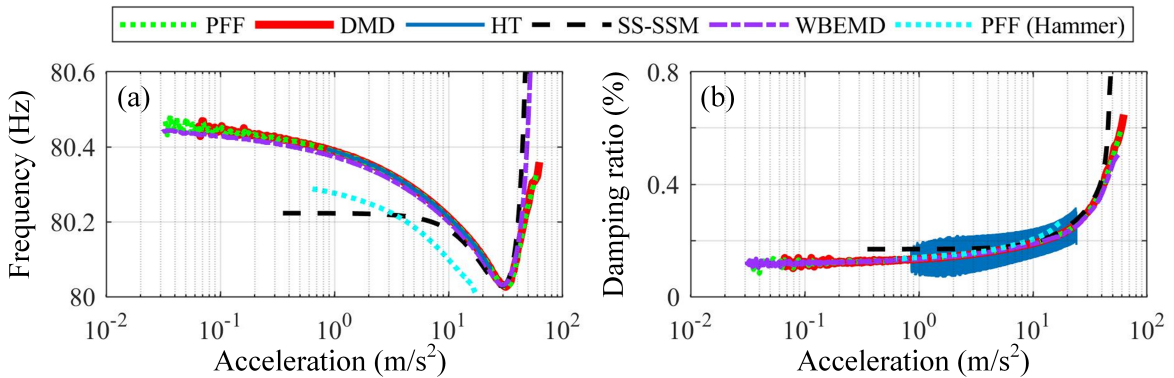


Figure 6: The identification results of the shaker ringdown testing by the PFF (green dotted line), DMD (red thick line), HT (blue thin line), WBEMD (purple dash-dotted line), and SS-SSM (black dashed line) methods, showing (a) backbone curve and (b) amplitude-damping curve. The cyan dotted lines are the identification results of the hammer testing using the PFF method.

345 One challenge for analysis is that most methods (e.g., the PFF method) are only capable  
 346 of dealing with a signal that contains exactly one frequency component. Thus, although

347 the free decay response of the shaker ringdown testing can be approximately regarded as  
 348 the signal with single frequency, the response is still filtered by the third-order butterworth  
 349 bandpass filter (that is used in the hammer testing) before the PFF method is applied.  
 350 The identification results of the HT method have much fewer data points than the other  
 351 methods, since the curves have been truncated to avoid the end effects [52]. Although the  
 352 identified frequency is smoothed [35] before being used to calculate the damping ratio, the  
 353 corresponding damping ratio shows the most noise among all of the methods. The results  
 354 obtained by the WBEMD method slightly differ from those of the others at high amplitudes.

355 The SS-SSM method, with respect to the other signal processing ones, analyzes the dy-  
 356 namics of a discrete-time state space model fit on the signals shown in **Fig. 5** (a). The results  
 357 in **Fig. 6** have been obtained using a 6-dimensional system with polynomial nonlinearities  
 358 up to order 5. The state space is made of measurement delays (using the typical settings  
 359 of autoregressive-moving-average models [34]), and the SSM expansion is then carried out  
 360 up to order 19. There is a discrepancy in the frequency of the SSM model and those of  
 361 the other backbones for low amplitudes since the SSM model has a well-defined linearized  
 362 frequency that, along with the model nonlinearities, is optimized in order to minimize the  
 363 model prediction errors.

364 Although **Fig. 4** and **Fig. 6** both show the results from free decay responses, these two  
 365 tests have their own characteristics. To further compare the results from the two tests,  
 366 the backbone and amplitude-damping curves obtained by the PFF method are presented  
 367 in **Fig. 6**. This figure shows that the highest amplitude in the identification results of the  
 368 shaker ringdown testing ( $80 \text{ m/s}^2$ ) is significantly larger than that of the hammer testing  
 369 ( $20 \text{ m/s}^2$ ). This shows that the shaker ringdown tests are able to acquire a wider range of  
 370 amplitudes than the hammer testing, providing more information about the variation of the  
 371 amplitude-dependent frequency and damping ratio. This is due to the broadband nature  
 372 of the hammer excitation in contrast with the possibility to isolate a single mode without  
 373 appreciably exciting the other modes with shaker ringdown testing. Specifically, the initial  
 374 acceleration in the hammer testing reaches  $500 \text{ m/s}^2$  (**Fig. 3** (a)), while the acceleration of  
 375 the shaker ringdown testing approaches only  $100 \text{ m/s}^2$  (**Fig. 5** (a)). However, the response  
 376 containing only the first mode has the initial acceleration of  $20 \text{ m/s}^2$  in the hammer testing  
 377 (**Fig. 3** (c)), and  $80 \text{ m/s}^2$  in the shaker ringdown testing (the zoom in **Fig. 5** (a)). Thus,  
 378 most of the energy is dissipated along a single mode or SSM in shaker ringdown testing, while  
 379 hammer testing allows the interplay of multi-modal nonlinear dissipation. As evidenced in  
 380 **Fig. 6** by the discrepancies between the amplitude properties, other frequency components  
 381 present in the signal influence the amplitude curves of the first mode, even though proper  
 382 filtering is performed. Therefore, if the aim of testing is to study modes in isolation, shaker  
 383 ringdown testing should be preferred due its capability to isolate single modes and to obtain  
 384 larger amplitude ranges.

### 385 *3.2. Forced-response Testing*

386 The forced-response testing aims at validating the previous free decay results. The  
 387 instantaneous frequency-amplitude curve of decaying vibrations has been often observed

388 to serve as a backbone for the frequency response for small forcing [46, 47, 43, 61, 62]  
 389 (in particular in [63, 64, 21] for non-conservative systems), and this relation is formally  
 390 established in [11]. Specifically, the backbone curve of a single mode SSM is the leading  
 391 order approximation of frequencies and amplitudes at which forced response maxima occur  
 392 for different forcing values.

393 In this research, both response-amplitude-control and force-amplitude-control stepped  
 394 sine testing are conducted on the HBRB for extracting forced responses. These are measured  
 395 by accelerometers to identify the backbone and amplitude-damping curves. The difference  
 396 between the two types of testing is that, in the response-amplitude-control testing, the re-  
 397 sponse amplitude of the driven point is controlled to be constant. By contrast, the amplitude  
 398 of the shaker force is controlled to remain fixed in the force-amplitude-control testing. Since  
 399 the first mode of the HBRB is concerned, the excitation frequency is swept from 79 Hz to  
 400 81 Hz with the step of 0.05 Hz. The response-amplitude-control testing is conducted with  
 401 amplitudes of 0.5 g, 2 g, 4 g, 5 g, 6 g, 7 g, and 8 g, and the force-amplitude-control testing  
 402 contains the force levels of 0.2 N, 0.5 N, 1.0 N, and 1.5 N. More detailed description of  
 403 the forced-response testing can be found in Part I [48]. **Figure 7** (a) shows the measured  
 404 response amplitude for the two types of forced testing, depicting the response-control testing  
 405 in red and in green the force-control testing. In what follows, the principal harmonics in  
 406 force and acceleration signals are only considered as higher order harmonics have negligible  
 influence for the system under analysis, cf. **Sect. 3.4.1**.

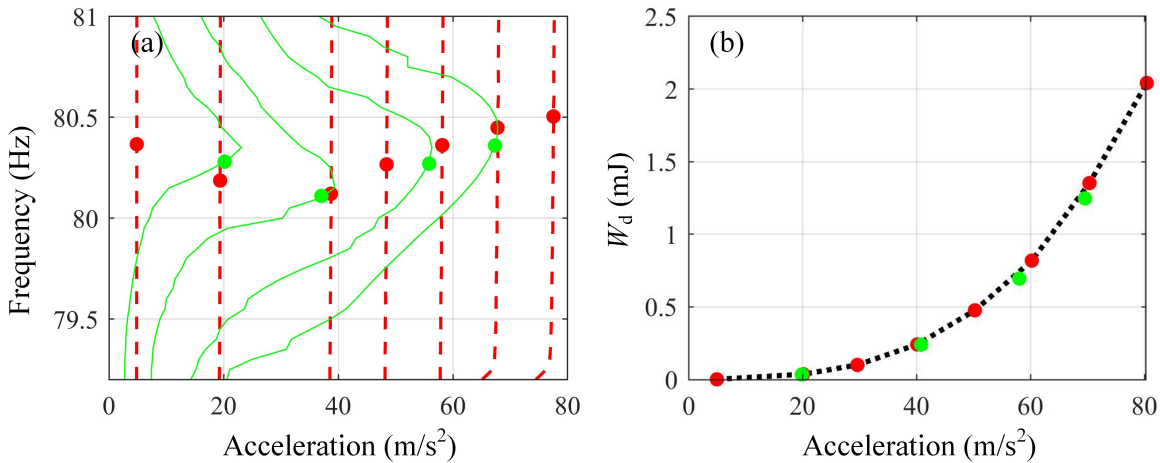


Figure 7: The results of the forced-response testing, showing (a) measured response amplitude in the response-control testing (red dashed line) and force-control testing (green dashed line) with the respective resonance frequency indicated with circles, and (b) dissipated energy over one oscillation cycle at resonance frequencies shown with consistent colors. Here, the black line connects red markers.

407  
 408 The extraction of maximal or resonant responses is performed by selecting the responses  
 409 satisfying the quadrature phase-lag criterion [46, 44, 61]. Here, the displacement response of  
 410 the driven point has approximately a  $90^\circ$  phase lag with respect to the forcing signal. The  
 411 identification of the backbone curve is then straightforward for both tests and is indicated by

412 circles in **Fig. 7** (a). The identification of the damping ratio, however, is not as immediate.  
 413 For a periodic trajectory of a mechanical system, the energy principle states that the work  
 414 done by non-conservative forces must be zero in one oscillation cycle [63, 44]. In other  
 415 words, the input energy provided by the shaker  $W_a$  is dissipated, mainly due to friction  
 416 occurring at the interface. **Figure 7** (b) shows the energy dissipated  $W_d$  in one cycle of  
 417 resonant oscillation. By denoting the first harmonic component of the acceleration of the  
 418 driven point response  $\ddot{q}(t) = -A \sin(\Omega t)$ , the forcing amplitude  $F$ , and the resonant forcing  
 419 frequency  $\Omega$ , it holds that

$$W_a = \int_0^{2\pi/\Omega} \dot{q}(t) F \cos(\Omega t) dt = \frac{\pi A F}{\Omega^2}, \quad (1)$$

420 and the dissipated energy over one oscillation cycle is

$$W_d = \int_0^{2\pi/\Omega} 2m^* \zeta_A^* \Omega \dot{q}^2(t) dt = 2m^* \zeta_A^* \frac{\pi A^2}{\Omega^2}, \quad (2)$$

421 where  $m^*$  is the equivalent modal mass. An equivalent, amplitude-dependent, modal damp-  
 422 ing ratio  $\zeta_A^*$  can be determined from the equality  $W_a = W_d$

$$\zeta_A^* = \frac{F}{2m^* A}. \quad (3)$$

423 The modal mass is estimated by calibration with the results from the free decay response,  
 424 which results in  $m^* = 1.55$  kg. Another computation of the dissipated energy is introduced  
 425 later in **Sect. 3.4.3** exploiting full-field data from DIC.

426 The comparison between amplitude properties extracted from forced testing and the ones  
 427 from free decay response are shown in **Fig. 8**. The results of the response-control and force-  
 428 control testing are close to each other. Additionally, the results of the two forced-response  
 429 testing are in good accordance with those of the shaker ringdown testing, especially in the  
 430 curves of damping. Although there is some discrepancy in the identification of frequency,  
 431 the trend of the backbones obtained by the forced-response testing is the same as that  
 432 obtained by the shaker ringdown testing. The partial discrepancy could be justified by the  
 433 low resolution (0.05 Hz) of forced testing, some control limitations at low amplitudes, and  
 434 the influence of the metallic stinger used to attach the beam to the shaker. The latter results  
 435 in an overall stiffness increase, consequently shifting the frequencies as shown in **Fig. 8** (a).  
 436 Moreover, from Eqs. 1-3, the ratio between the external forcing amplitudes  $F$  and those of  
 437 the inertial forces  $m^* A$  is equal to  $2\zeta_A^*$ , which stays small, thereby justifying the comparison  
 438 between the results obtained via free decay response and forced response.

439 Although the response-control and force-control stepped sine testing are both capable of  
 440 identifying nonlinear characteristics with high accuracy, they have several drawbacks. First,  
 441 the identification is generally sparse as forced experiments require a considerable amount of  
 442 time. Additionally, at least for the system under investigation, the experimental equipment  
 443 showed limitations in reaching low amplitude levels of excitation.

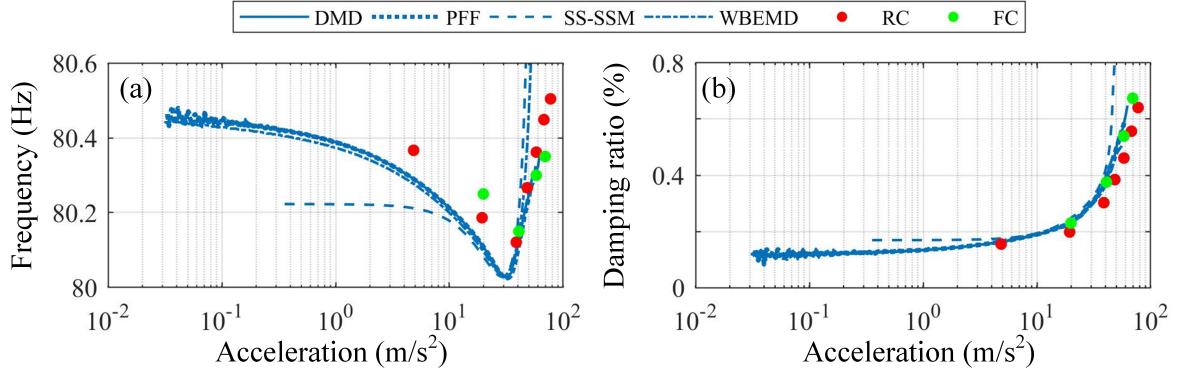


Figure 8: The identification results of the response-control testing (red circles), force-control testing (green circles), and shaker ringdown testing (blue lines) using different identification methods, showing (a) backbone curve and (b) amplitude-damping curve.

### 444 3.3. Free Decay Response by DIC

445 Following the analysis of the accelerometer data in the previous section, this section  
 446 focuses on the measurements from the two high-speed cameras, which are used to capture  
 447 the displacement of the HBRB during the shaker ringdown testing. DIC [40, 41] is applied to  
 448 the recorded images to calculate the displacement responses. Since there are multiple points  
 449 in the images, the raw response and the filtered response (filtered by the same butterworth  
 450 filter that is used in the hammer testing) of one of the points are selected to be shown  
 451 in **Fig. 9**. The raw response mainly contains the first mode of the HBRB. Although the  
 452 decaying signals shown in **Fig. 5** (a) and **Fig. 9** (a) comes from the same experiment,  
 453 the presence of other frequency components is more evident in **Fig. 9**. In particular, the  
 454 oscillation due to sway mode of the beam, at low frequency, is clearly visible, especially at the  
 455 initial time instances of the measurement. The reason why it is more visible in the DIC data  
 456 of **Fig. 9** (a) than in the accelerations in **Fig. 5** (a) is that DIC processes displacements,  
 457 which are more evident at low frequencies.

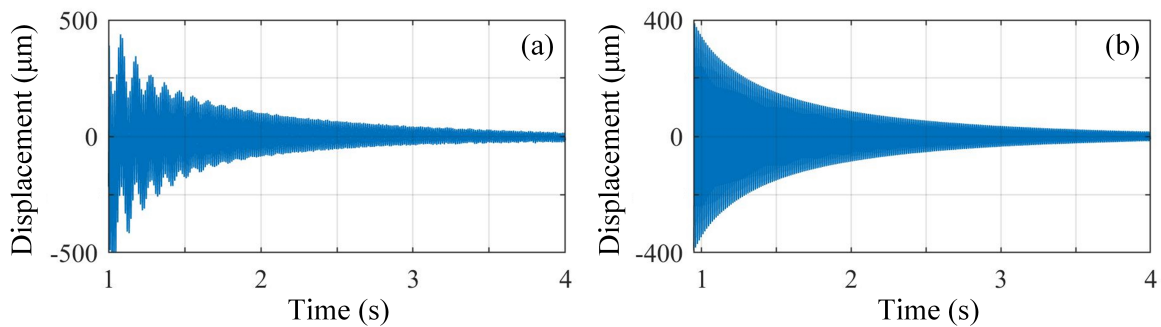


Figure 9: The displacement response of one of the points of the HBRB in the shaker ringdown testing, showing (a) raw response obtained by DIC and (b) filtered response.

458 Since all of the methods have already been compared when dealing with the acceleration

459 signal in the hammer testing and shaker ringdown testing in **Sect. 3.1**, only the PFF  
 460 and DMD methods are applied here to the displacement responses obtained via DIC. An  
 461 important distinction between the two methods is that the PFF method is applicable to  
 462 the response signal of one single point, while the DMD method is based on the Hankel  
 463 matrix and can thus be applied to the response signals of all of the points together. In  
 464 other words, given a large number of displacement responses, the PFF method will provide  
 465 multiple curves, each of which corresponds to one of the points, and the DMD method will  
 466 result in only one curve that contains the information of all of the points.

467 Note that the time series obtained from DIC have different signal-to-noise ratios across  
 468 the length of the beam. In particular, time series whose locations lie close to the nodes  
 469 of the mode shape are predominantly affected by noise. Additionally, during a free decay,  
 470 signal-to-noise ratios worsen as time increases so that the time series have to be truncated  
 471 to preserve a high signal-to-noise ratio.

472 The traditional way to present the backbone curve of a structure is to plot the fre-  
 473 quency as a function of the response amplitude. **Figure 10** shows the identification results  
 474 of the PFF method regarding the backbone curves and amplitude-damping curves of the  
 475 measurement points on the beam.

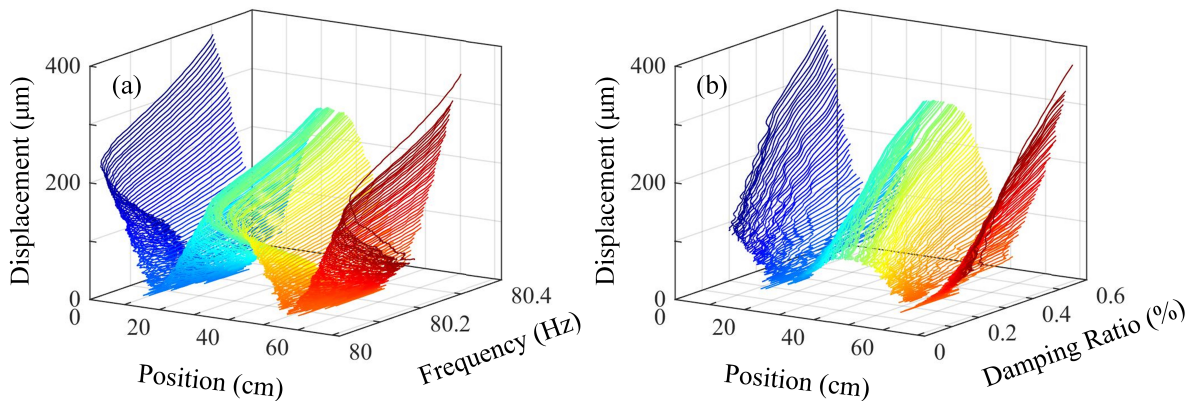


Figure 10: The identification results of the DIC data using the PFF method, which correspond to the measurement points at different positions, showing (a) backbone curves and (b) amplitude-damping curves.

476 It can be seen from **Fig. 10** that although the range of the frequency/damping ratio  
 477 is the same for each point, the range of the displacement amplitude differs. Since all of  
 478 the points are in different positions on the beam, the amplitude of each point is different  
 479 from each other. Thus, it may not be particularly meaningful to express the frequency  
 480 or damping ratio as a function of the displacement amplitude. Another way to normalize  
 481 the response is to plot the frequency and damping ratios as a function of time, i.e., the  
 482 instantaneous frequency and instantaneous damping ratio. The results of both the PFF  
 483 and DMD methods are shown in **Fig. 11**, whose x-axes appear reversed when compared to  
 484 amplitude-based plots, cf. **Fig. 6**.

485 Regarding the displacement responses obtained by DIC, both the PFF method and the



486 DMD method bring about similar results of the instantaneous frequency and damping ratio.  
 487 As the PFF method is applied to the response signal of each measurement point, it provides  
 488 a large number of curves whose variability is recast to noise effects. By contrast, because the  
 489 DMD method combines the response signals of all of the points together and calculates one  
 490 curve for each amplitude-dependent characteristic, the results of the DMD method can be  
 491 regarded as the average of those obtained by the PFF method. Moreover, the instantaneous  
 492 frequency and damping ratios from DIC data are consistent with those from accelerometer,  
 493 as show in **Fig. 11** which includes the curves obtained via DMD from accelerometer data.  
 494 However, acceleration signals have longer time histories than those obtained by the DIC data  
 495 due the memory of the cameras. In particular, videography-based techniques are limited by  
 496 the trade-off between spatial resolution and sampling frequency, which both determine the  
 497 maximum length of the time signals possible based on available camera memory.

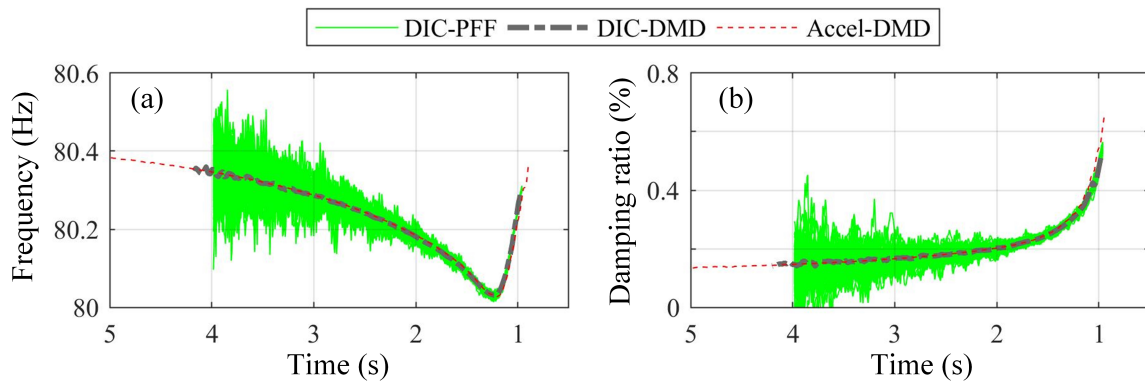


Figure 11: The results obtained by the DIC data and acceleration data using PFF (green solid lines) and DMD (grey dashed-dotted line for DIC data and red dashed line for acceleration data) methods, showing (a) instantaneous frequency and (b) instantaneous damping ratio.

### 498 3.4. Additional Information Brought by DIC

499 Unlike the accelerometers that are sparsely distributed on the beam, the cameras are  
 500 able to measure the displacement of an entire face of the HBRB, so that all of the points  
 501 on the beam are recorded simultaneously. After the images are processed by DIC, the full-  
 502 field information of the beam is available (i.e., the displacement responses of the points are  
 503 obtained). Hence, DIC data allows the detailed analysis of the beam's mode shape and also  
 504 the evaluation of its energy. Note that a similar degree of information cannot be achieved  
 505 by accelerometers since it is difficult to place dense accelerometers on the structure without  
 506 changing its characteristics at the same time [15].

#### 507 3.4.1. Measurements of Mode Shapes from Ringdown Testing

508 In shaker ringdown testing, the motion of the beam during the free decay can be assumed  
 509 as an estimation of the mode shape of the considered mode, since the beam is released from  
 510 the quadrature between force and acceleration of the driven point [48]. The dominance of the

511 fundamental harmonic during the decay is shown in **Fig. 5** (b) and the motion synchronism  
 512 is further discussed and verified both at the end of this subsection and in **Fig. 13**. The mode  
 513 shape is calculated as the deflection of the beam at each maximum of the cycle of motion.  
 514 In this way, the time-varying (and thus, amplitude-varying) mode shape is obtained. The  
 515 number of measurement points depends on the spatial resolution chosen for DIC [40, 41]  
 516 [48]. The ultimate result consists of two lines describing the mode shape of the upper and  
 517 lower part of the beam. These lines are displayed on a sketch of the beam in **Fig. 12** (a).

518 In **Fig. 12**, the shape of the first mode over 3.5 s of free decay is shown. This time  
 519 duration is approximately 277 cycles of vibration; for visualization purposes, only 30 time  
 520 instants are presented in **Fig. 12**. The abscissa is the position of the measurement points and  
 521 the ordinate (Y axis) shows the amplitude of the responses of the corresponding measurement  
 522 points. Each line consists of 206 DIC data points. It can be seen from **Fig. 12** that the  
 523 amplitudes of the mode shapes reach zero at around 17 cm and 55 cm, which are the two  
 524 positions of the node points of the first mode. Additionally, there is a discontinuity of the  
 525 curvature of the mode shape at 29.7 cm for the lower line (blue) and at 42.3 cm for the upper  
 526 line (red). These locations correspond to the left and right edges of the interface between  
 527 the two half beams, shown in **Fig. 12** (a). When the amplitude of motion increases, the  
 528 described discontinuity intensifies, due to the geometry of the beam. The mode shapes in  
 529 **Fig. 12** refer only to the maxima of the cycle of motion, i.e., a deflection downwards of  
 530 the center of the beam and upwards of the extremities. By contrast, the nodes of the mode  
 531 shape preserve their location when motion amplitude changes.

532 The discontinuity in the mode shape is due to the opening of the interface on the lower  
 533 edge (**Fig. 12** (c)), and closing of the interface on the upper edge (**Fig. 12** (d)). This  
 534 behavior is also visible in **Fig. 12** (b), which reconstructs the maximum modal displacement  
 535 of the entire beam from the two lines of that come from the DIC processing (**Fig. 12** (a)),  
 536 and is consistent with other measurements of a similar lap joint [65, 66]. The difference  
 537 between the upper and lower lines shows that the maximum and minimum differences occur  
 538 at the locations of the discontinuities (29.7 cm and 42.3 cm (**Fig. 12** (e))), clearly reflecting  
 539 the gap opening at the interface. However, the deflection ratio between upper and lower  
 540 lines, depicted in **Fig. 12** (f), does not reveal significant variations over time.

541 Full-field data can also be used to assess whether the decaying vibrations shows synchro-  
 542 nism or not. According to the original definition of a conservative nonlinear normal mode by  
 543 Rosenberg [67], a periodic motion is synchronous if all maxima and minima of the position  
 544 time series are achieved at the same time. Due to the presence of damping, synchronous  
 545 motion is not guaranteed, but resonant motions in weakly damped structures have been still  
 546 observed to be (approximately at least) synchronous [46, 47]. In **Fig. 13** (a), a time-extract  
 547 of the full displacement field for the lower line of the beam is illustrated, where the abscissa  
 548 displays time, cf. **Fig. 7** (b), the ordinate displays the position along the beam, and the  
 549 color scale depicts the displacement magnitude. The horizontal white lines that appear from  
 550 the colormap correspond to the nodes of the mode shape, while the vertical white lines show  
 551 the times at which all of the responses are equal to zero. At least for this time portion, the  
 552 decaying motion may already appear synchronous. This fact has also been verified in detail

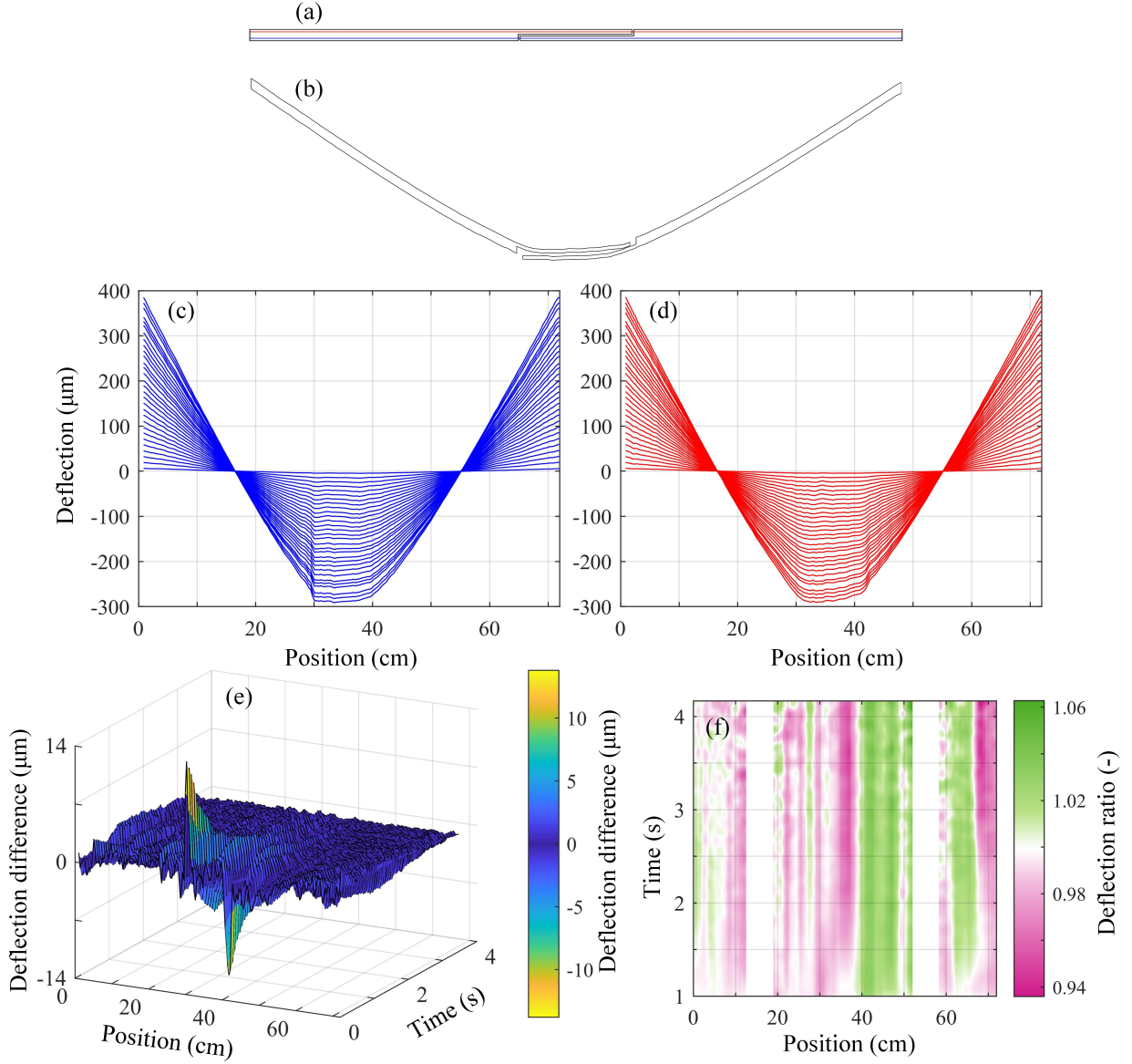


Figure 12: Amplitude-dependent mode shape of the 1<sup>st</sup> mode of the HBRB from shaker ringdown testing, showing (a) sketch of the HBRB with the two lines identified with DIC, (b) maximum deflection shape, (c) DIC lower line, (d) DIC upper line, (e) difference, and (f) ratio between upper and lower line.

553 exploiting the PFF method, which computes the time at which a local extremum (either  
 554 maximum or minimum) occurs for every displacement time series. Let  $t_{ext}(l, k)$  be the time  
 555 at which the  $k$ -th extremum occurs at the location  $l$  and, by calling  $\Omega$  the location domain,  
 556 a metric to describe the synchronous behavior of the system is defined as

$$\Delta S(k) = \max_{l \in \Omega} |t_{ext}(l, k) - \bar{t}_{ext}(k)|, \quad \bar{t}_{ext}(k) = \frac{1}{|\Omega|} \sum_{l \in \Omega} t_{ext}(l, k). \quad (4)$$

557 The quantity  $\Delta S(k)$  acts as a metric for assessing the degree of synchronism in the structure  
 558 and it is plotted in **Fig. 13** (b) once normalized by the sampling time  $t_{samp} = 0.0002$  s.  
 559 For computing  $\Delta S(k)$ , the points close to mode shape nodes have been excluded from the  
 560 calculation domain  $\Omega$  since their displacement time series are dominated by noise. Moreover,  
 561 the increasing variability of  $\Delta S$  over time is due to the deterioration of the signal-to-noise  
 562 ratio for decreasing motion amplitudes. This analysis leads to the conclusion that the time  
 563 at which an extremum is detected for a particular location differs by less than one sampling  
 564 interval from the average time at which the same extremum is detected at any other location  
 565 across the beam. A similar result holds for the times at which the displacement field is zero.  
 566 Hence, the decaying vibration can be approximated as synchronous and this evaluation has  
 567 crucial consequences for the evaluation of the energy and the mode shape of the beam, as  
 568 discussed in **Sect. 3.4.3**.

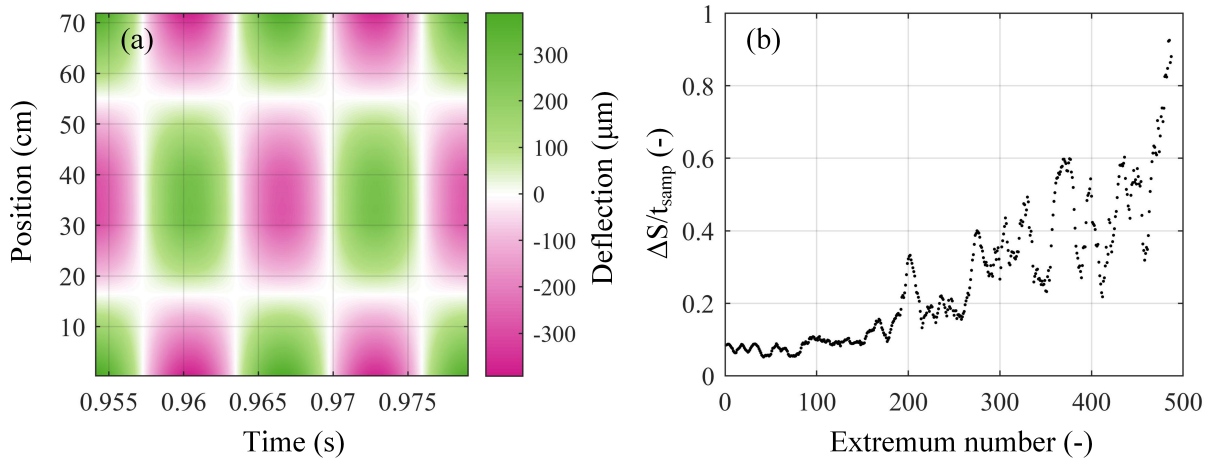


Figure 13: Assessment of motion synchronism, showing (a) full-field motion of the beam lower line for a selected time interval and (b) variability of the peak time identified via the PFF method.

### 569 3.4.2. Measurements of Mode Shapes from Response-Amplitude-Controlled Fixed Sine Test- 570 ing

571 Further information of the beam's mode shape can be extracted from response-amplitude-  
 572 controlled fixed sine testing. The mode shape can be estimated from the time series, which  
 573 has a controlled frequency, unlike the decaying vibrations measured in the ringdown tests.  
 574 As the fixed sine testing is conducted to ensure that the system achieves a steady state  
 575 response, multiple forcing cycles of the steady state response are able to be measured for  
 576 each excitation amplitude. Tests are conducted for the response levels 0.5 g, 2 g, 4 g, 8 g,  
 577 and 10 g, and the shaker is tuned for each amplitude to excite the first mode, as described  
 578 in [48]. The same considerations of **Sect. 3.4.1** about the synchronism of the motion apply  
 579 here. Moreover, **Fig. 14** shows the response spectrum of the beam extremity for the highest  
 580 level of the excitation, where the higher harmonics are larger. The peaks below 8 Hz are  
 581 rigid body modes, and the peak at 146 Hz is due to the fan internal to each of the high-speed

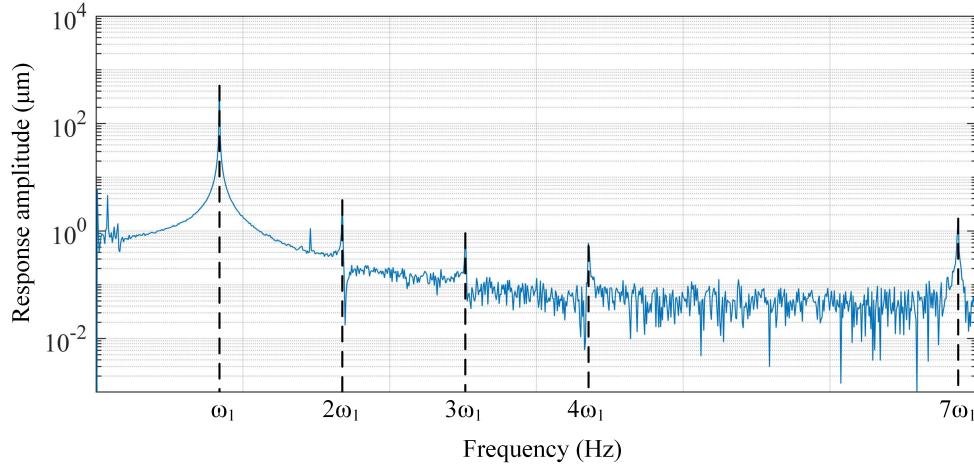


Figure 14: Response spectrum of response-amplitude-controlled fixed sine test at 10 g.

582 cameras [48]. Although several harmonics of the first mode are present, their amplitude is  
 583 several orders of magnitude lower than the fundamental harmonic. In order to smooth the  
 584 results, the data are filtered by a third-order bandpass butterworth filter centered at the  
 585 first natural frequency with a bandwidth of 40% of the first natural frequency. The results  
 586 of the mode shape estimation are shown in **Fig. 15** for each amplitude level, both for the  
 587 upper and lower lines. Two additional plots shed further light on the amplitude-dependent  
 588 properties of the mode shape. The first one, **Fig. 16** (a), shows the upper line mode shape  
 589 of a central portion of the beam, normalized with respect to the left tip displacement. From  
 590 the lowest to the highest excitation amplitude, the displacements of the central part of the

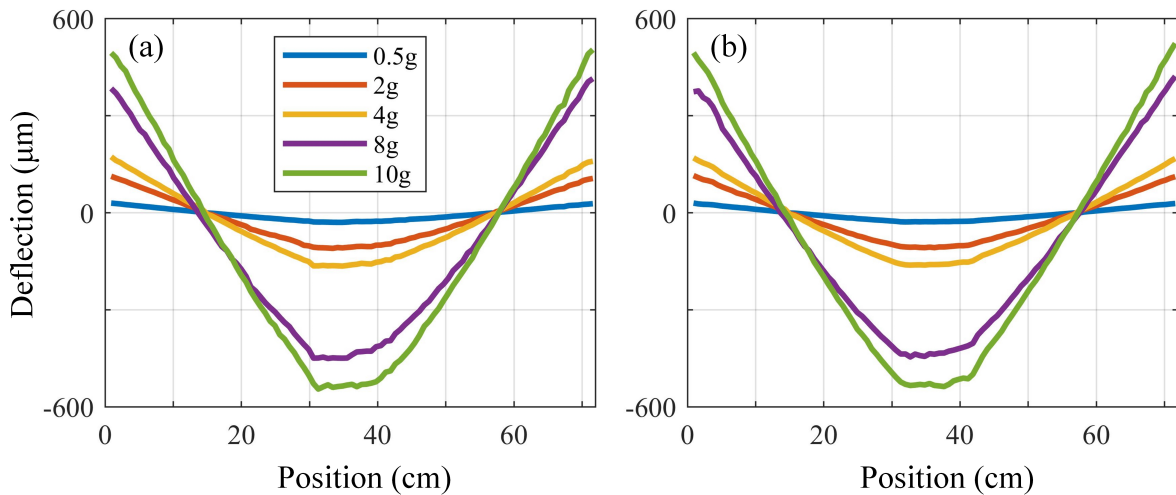


Figure 15: Amplitude-dependent mode shape of the 1<sup>st</sup> mode of the HBRB from response-amplitude-controlled testing, showing (a) DIC lower line and (b) DIC upper line. Each line is the mode shape calculated from the maximum displacement for an excitation level.

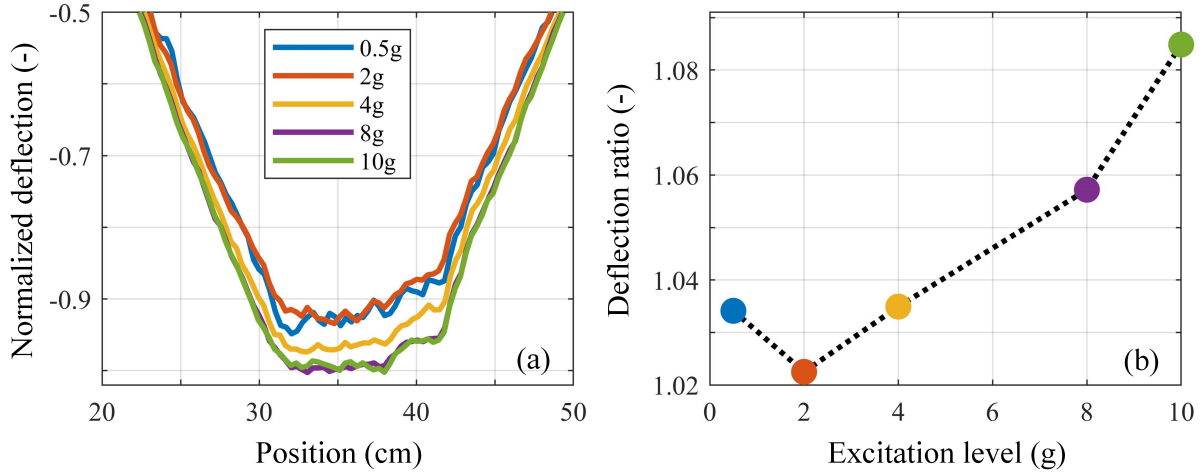


Figure 16: Amplitude-dependent mode shape of the 1<sup>st</sup> mode of the HBRB from response-amplitude-controlled testing of DIC upper line, showing (a) normalized deflection and (b) deflection ratio between the middle of the beam to the tip displacement.

591 beam increase approximately by 7%, with respect to the tip displacement, but decrease for  
 592 the 2 g level. This suggests a softening behavior of the modal stiffness. The overall trend  
 593 resembles that of the backbone curve, but the variation in this case is greater with respect  
 594 to the frequency change in the backbone curve, cf. **Fig. 8**, since this is a local phenomenon  
 595 while the natural frequency depends on global characteristics. This behavior is illustrated  
 596 more compactly in **Fig. 16** (b) with the deflection ratio between the displacement of the  
 597 middle of the beam to the tip displacement. Precisely, the tip displacement is computed as  
 598 the average between the displacements of the ten leftmost points from DIC, and the middle  
 599 displacement is the average among the ten displacements around the beam center. While the  
 600 locations of the mode shape nodes remain unchanged, the principal variation of the mode  
 601 shape is concentrated in the middle of the beam, suggesting that further studies should  
 602 connect this macroscopic observation with microscopic phenomena of the contact patch.

### 603 3.4.3. Energy of the Beam

604 The total energy of a structure consists of kinetic energy and potential (strain) energy.  
 605 As shown in the STFT analysis (**Fig. 5** (b)), the free decay response obtained from the  
 606 shaker ringdown testing only contains the first mode of the HBRB. The potential energy is  
 607 assumed to be zero at the equilibrium position, corresponding to the configuration when all  
 608 displacements are zero. As verified previously, the vibration is synchronous and, during the  
 609 decay, every point passes through zero at the same time. At these time instances, the total  
 610 energy of the structure is therefore equal to the kinetic energy.

611 As DIC provides responses of a large number of points, the HBRB can be discretized into  
 612 lumped mass elements. The filtered displacement responses (**Fig. 9** (b)) can be transferred  
 613 into velocity responses by numerical differentiation. Thus, with the full-field information of

614 the structure and the mass of the whole beam ( $M = 1.796$  kg as measured), the energy  $E$   
615 at time instant  $t$  can be approximated by

$$E(t) = \sum_{i=1}^N \frac{1}{2} m v_i^2(t), \quad (5)$$

616 where  $N$  is the number of the measurement points,  $v_i$  is the velocity of the  $i$ th point, and  
617  $m = M/N$  is the mass of each measurement point.

618 Here, the total number of the measurement points is  $N = 206$ , which is determined during  
619 the processing of the DIC. The energy trend, computed using all measurement points, over  
620 the shaker ringdown testing is shown in **Fig. 17** (a) with a black line, and the red dots depict  
621 energy peaks. In order to study the influence of the number of the measurement points on  
622 the calculation of  $E$ ,  $N$  is varied from 5 to 206 over an equally spaced grid. **Figure 17**  
623 (b) shows the energy of the beam, at the time instant of the first peak, as a function of  $N$ ,  
624 which converges for  $N > 100$ . In other words, the number of points  $N$  needed to have an  
625 accurate estimation of the energy is higher than 100. This spatial resolution is not achievable  
626 with contact sensors such as accelerometers or strain gauges, but only with a non-contact  
627 measurement approach, such as videography and DIC.

628 The relationships between energy and displacement/acceleration amplitudes for the ac-  
629 celerometer located on the shaker (see Accel 3 in **Fig. 2** (a)) is shown in **Fig. 17** (b). A  
630 clear, quadratic trend, as defined in Eq. (5), is observed. The accuracy of the energy cal-  
631 culation by DIC is validated in **Fig. 17** (d) by means of the dissipated energy. The red  
632 and green dots are the work done by the forcing for the response-control (red) and force-  
633 control (green) testing procedures for different amplitudes, cf. **Fig. 7** (b). These quantities  
634 are computed using the measurement of the shaker force and the acceleration of the driven  
635 point. As argued in **Sect. 3.2**, this work is equal to that done by the non-conservative  
636 forces over one oscillation cycle. For the shaker ringdown, the peaks in the kinetic energy  
637 correspond to the total energy of the system, since the motion is synchronous. Hence, the  
638 difference between the energy of the  $k$ -th peak and that of the  $(k+2)$ -th peak (see the inset  
639 in **Fig. 17** (a)) is the work done by the non-conservative forces in one oscillation cycle. The  
640 dissipative work computed from forced testing and from shaker ringdown is expected to be  
641 comparable from theoretical speculations. Indeed, **Fig. 17** (d) shows that there is a very  
642 good agreement with the two. Here, the blue line shows the dissipative work, computed  
643 using the kinetic energy decay obtained from full-field data, plotted against the acceleration  
644 amplitude level. It agrees well with the energy measurements obtained in forced testing  
645 using a single accelerometer.

646 The energy computed here could serve for model validation or tuning, but it is also useful  
647 to construct frequency-energy plots, typically adopted in the NNM literature [9] to charac-  
648 terize the dynamical feature of nonlinear responses. Frequency-energy plots are particularly  
649 meaningful for conservative systems, where each NNM orbit lies on a specific energy level  
650 [29]. However, they are relevant also for weakly forced, damped structures as their resonant  
651 periodic motions typically show small deviations from their conservative counterpart, which

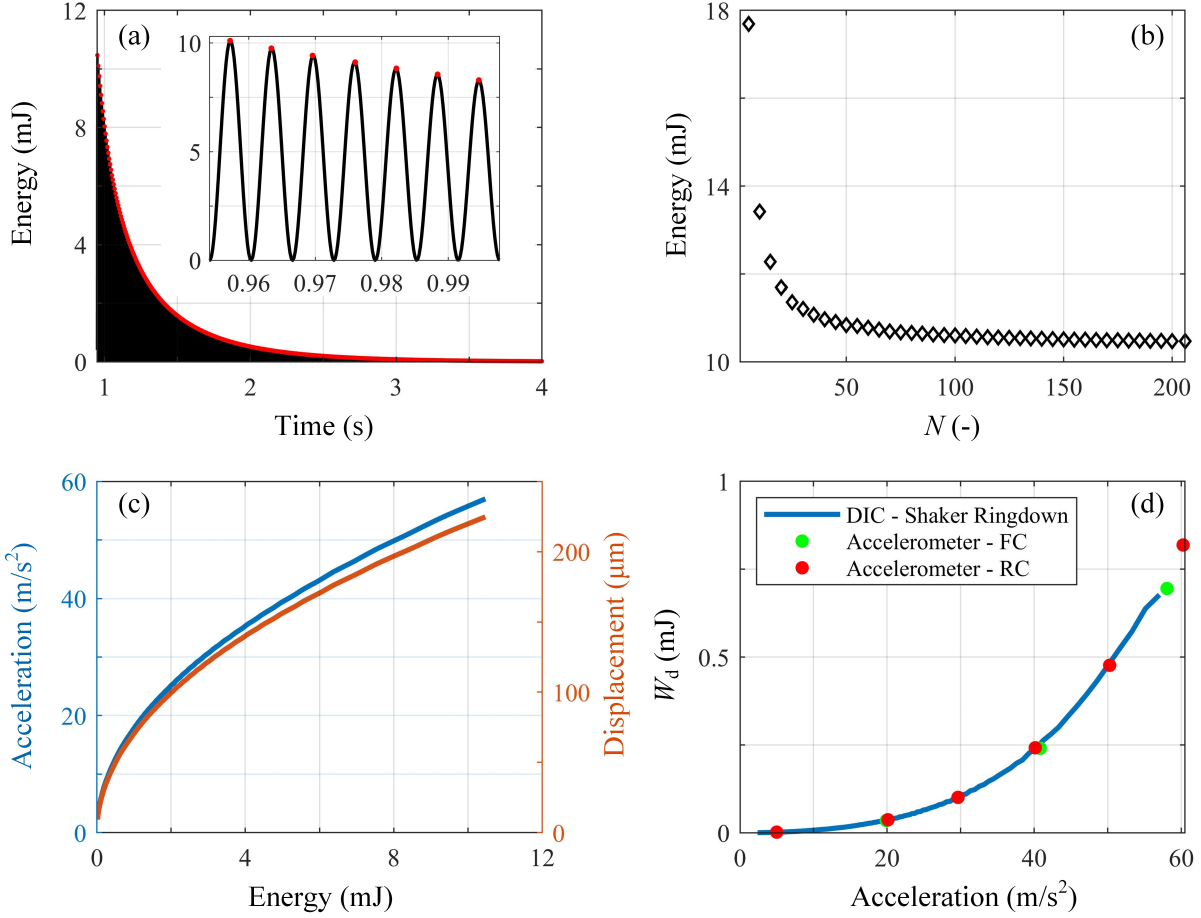


Figure 17: The energy of the HBRB where (a) shows the trend of the kinetic energy over time (with the inset showing the first seven peaks) and (b) the dependence of the energy on the number of the measurement points. Plot (c) shows the relation between energy and displacement/acceleration amplitudes at the location of Accel 3 in **Fig. 2**. The last plot in (d) shows the dissipated energy over one oscillation cycle computed via accelerometer measurements from forced testing (red and green dots, cf. **Sect. 3.2**) and the same quantity extracted using full-field data from shaker ringdown.

652 can be neglected [46, 47, 61]. Experimentally, the construction of frequency-energy plots is  
 653 not practical since measuring the energy of a structure accurately is impractical with ac-  
 654 celerometers, and thus the frequency-amplitude (typically acceleration amplitude) backbone  
 655 plots are often reported instead, which is a flawed metric [19, 20]. In this study, however,  
 656 the calculation of energy is feasible due to the full-field information from DIC.

657 **Figure 18** shows the nonlinear characteristics of the system using the full-field DIC  
 658 data. Here, all of the 206 DIC points are considered to ensure a sufficiently converged result,  
 659 since the use of  $N = 206$  instead of a smaller value, such as 100, comes at no significant  
 660 computational cost increase. As an alternative to the displacement or acceleration amplitude  
 661 used previously (see **Figs. 8** or **11**), the damping and frequency are now combined with  
 662 the energy calculations, as shown in **Fig. 18** (a)-(d). Specifically, **Fig. 18** (c) and (d)



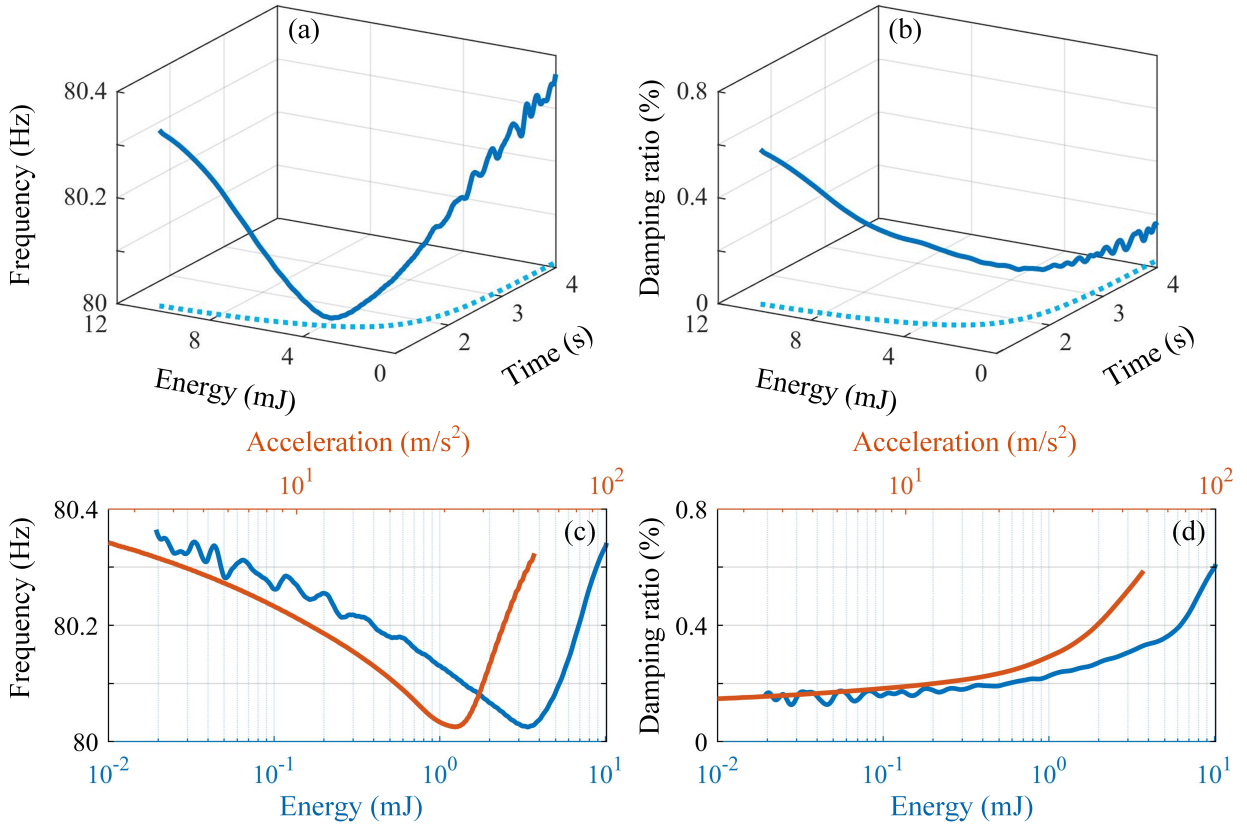


Figure 18: The relationship between the time, energy, amplitude, and nonlinear characteristics of the HBRB, including (a) relationship between time, energy, and frequency, (b) relationship between time, energy, and damping ratio, (c) energy-dependent and amplitude-dependent frequencies, (d) energy-dependent and amplitude-dependent damping ratios. The latter amplitudes refers to the acceleration amplitudes measured by Accel 3 in **Fig. 2**.

663 superimpose amplitude-dependence and energy-dependence. Clearly, the two properties  
 664 show consistent values of frequency and damping whose amplitudes are related by a scaling  
 665 factor and a quadratic relationship. Amplitude-dependent property plots act as proxies for  
 666 the energy-dependent property plots in this case, but the latter may provide a more global  
 667 perspective on the system under analysis. However, the energy-dependent plots in **Fig. 18**  
 668 (c) and (d) show an increased variability towards small amplitudes, due to the degradation  
 669 of the signal-to-noise ratio in DIC measurements. By contrast, amplitude-dependent curve  
 670 shows a smoother behavior since the accelerometer data, at the same amplitudes, is not  
 671 subject to these noise issues. It is important to note that if the system was not vibrating  
 672 synchronously, the amplitude dependent plots may have a less clear relationship with the  
 673 energy plots depending on which point is used to measure the amplitude.

#### 674 4. Comparison of the Methods from a Practitioner Viewpoint

675 Nonlinear system identification, as shown in this paper, requires a toolbox approach  
676 in which different techniques have to be deployed depending on the system under analysis  
677 and the testing procedure. Force-response testing tends to provide robust information since  
678 steady states can be properly isolated, but the shaker unavoidably influences the system  
679 dynamics. Hammer excitation is easy to perform, but it generates a multi-modal component.  
680 This is less than ideal if the identification targets a particular mode of the structure, as much  
681 of the energy is distributed across modes that are not of interest. In this context, shaker  
682 ringdown should be the preferred test method as it combines the advantages from both  
683 shaker testing (i.e., single frequency and high level of excitation at that frequency) and  
684 hammer excitation (i.e., transient free decay data). However, decoupling the shaker from  
685 the structure is often not a straightforward task [48].

686 Excluding SS-SSM, all of the methods used in this paper are signal processing techniques  
687 suited for time-frequency analyses. The extraction of frequency patterns in a time series  
688 varies according to the theoretical foundations of the method, but a common requirement of  
689 time-frequency methods is the need to filter the transient data before extracting the nonlinear  
690 characteristics. By contrast, for SS-SSM, the frequency-amplitude relation is intrinsic in the  
691 identified nonlinear model of the system. Encouragingly, all of the techniques showed good  
692 agreement in the results for shaker ringdown and hammer testing.

693 Among all of the methods, DMD is the only one that was originally born for the analysis  
694 of multi-point measurements of the dynamical system. The other techniques are more suited  
695 for single-point measurements, and hence required suitable rearrangements in order to be  
696 applied to multiple measurements simultaneously. Moreover, the HT, PFF, SS-SSM, and  
697 WBEMD methods have been developed and extensively used for studying transient regimes  
698 in mechanical systems. By contrast, DMD, broadly known in the fluid dynamics commu-  
699 nity for unfolding the dynamics of ergodic attractors, has presented challenges for tackling  
700 transient phenomena [68]. However, the results of DMD show a very good agreement with  
701 the other techniques in the context of this research.

702 Every technique deployed in this paper presents either computational or implementation  
703 challenges. From the complexity viewpoint, all of the techniques have drawbacks. The HT  
704 and PFF methods require filtering that needs to be manually tuned for the best results [58]  
705 in order to retain the core harmonic information of the time series [25]. Moreover, the HT  
706 method suffers from the issue of end effect in the estimation of the damping curve, even  
707 though appropriate signal pre-processing is performed, cf. **Fig. 4** (b) and see [25]. For the  
708 SS-SSM method, model fitting is fundamental for obtaining adequate results [34], which  
709 requires a computational routine to determine the coefficients that describe the SSM and its  
710 dynamics [31]. For DMD, the results depend on the selection of time scales (e.g., sampling,  
711 windowing) and of the number of dynamic modes. Finally, WBEMD needs an optimization  
712 in order to identify the intrinsic mode functions, and hence it requires some application  
713 dependent parameters [57]. Overall, DMD, HT, and PFF may seem to be less complex  
714 in terms of implementation for the context of the paper. However, SS-SSM may also be

715 exploited to construct data-driven reduced models, and WBEMD can efficiently uncover  
716 multi-modal transient phenomena.

## 717 5. Conclusions

718 This paper compared the performance of multiple nonlinear identification methods based  
719 on the experimental data from a jointed structure. The backbone curve and amplitude-  
720 damping curve were extracted from the signals using Hilbert Transform (HT), Peak Finding  
721 and Fitting (PFF), Dynamic Mode Decomposition (DMD), State-Space Spectral Submani-  
722 fold (SS-SSM), and Wavelet-Bounded Empirical Mode Decomposition (WBEMD). The re-  
723 sponses were measured by accelerometers in the experiments including hammer testing,  
724 shaker ringdown testing, and response/force-control stepped sine testing. During the shaker  
725 ringdown testing, apart from the accelerometers, two high-speed cameras recorded the im-  
726 ages of the jointed beam. Then, Digital Image Correlation (DIC) was adopted to extract  
727 the displacement responses from the images to reconstruct the mode shapes of the whole  
728 beam. The main conclusions of this paper are summarized as follows:

- 729 • Backbone and damping curves obtained from hammer testing and shaker ringdown  
730 testing showed a similar trend. However, the amplitude range in the shaker ringdown  
731 testing was wider than that in the hammer testing despite the peak amplitude of the  
732 acceleration of the tip of the beam being smaller in the shaker tests. This happens  
733 because the hammer testing excites multiple modes, while the shaker ringdown testing  
734 is able to excite one single mode of the structure. Especially at high amplitudes,  
735 the amplitude-dependent properties obtained via hammer testing deviated from those  
736 obtained via shaker ringdown testing, although proper filtering was applied. This  
737 observation proves that other modes can indeed influence the amplitude-dependent  
738 properties of a specific mode. Therefore, hammer testing should be carefully handled  
739 when aiming to extract single-mode nonlinear information.
- 740 • Amplitude-dependent properties computed from shaker ringdown testing and from  
741 forced-response testing presented good agreement, as expected from theoretical re-  
742 sults. In this paper, an energy criterion was used to extract backbone and damping  
743 curves from response/forced-control testing. Although it requires precise tuning and an  
744 effective decoupling mechanism, shaker ringdown testing is more efficient with respect  
745 to forced testing, where the acquisition of dense datasets can necessitate extensive  
746 experimental campaigns.
- 747 • Other than amplitude-dependent properties, which show good agreement with those  
748 coming from accelerometer data, full-field data from two high-speed cameras are able  
749 to reconstruct the mode shape of the beam. The deformation induced by the lap joint  
750 is visible and the variation of the mode shape with the amplitude of motion can be  
751 assessed as well as motion synchronism. Moreover, full-field data allows the definition  
752 of the kinetic energy of the beam, which is able to determine amplitude-dependent

753 properties. The value of the kinetic energy also showed good convergence properties  
754 as the number of measurement points was increased. Moreover, the dissipated energy  
755 computed from the decay of kinetic energy during a shaker ringdown test is consistent  
756 with the dissipated energy computed in forced testing. The metric of energy, which  
757 cannot be achieved with traditional accelerometers alone due to the need for a spatially  
758 dense data set for adequate convergence, can be useful for practices such as model  
759 validation and model updating. However, especially when dealing with long structures  
760 in free vibrations, the cameras are limited in memory, and thus they are only able to  
761 record a short period of time.

- 762 • All of the methods used in the paper agreed well in the extraction of amplitude-  
763 dependent properties from free vibrations. Other than showing performances, this  
764 paper also critically compared these methods, pointing out benefits and potential  
765 challenges in order to enrich the toolbox of practitioners in structural dynamics.
- 766 • In terms of the frequency-amplitude and damping-amplitude curves, the beam un-  
767 der investigation showed limited frequency change, i.e., 0.5 Hz, while the damping  
768 ratio rose from about 0.1% at low amplitude to more than 0.6% at high amplitudes.  
769 Additionally, the mode shape presented substantial amplitude dependence, especially  
770 in the area of the contact patch. The deflection ratio, which tries to quantify this  
771 variation, exhibited a similar trend to that of the backbone curve, but with a much  
772 wider range amounting to 7%. Thus, the strongest nonlinearities for the jointed beam  
773 studied were observed in the damping behavior and the variation of the mode shape.  
774 Further studies can help to uncover the connection of these macroscopic phenomena  
775 with interactions occurring in the contact patch.
- 776 • The relationship between the energy and nonlinear frequency and damping ratio of  
777 the beam have been obtained due to the full-field DIC data. The energy-dependent  
778 characteristics provide the information of the whole beam instead of a few points  
779 on the structure. Accelerometers, as a traditional way of measurement, make the  
780 measurement of the beam's energy prohibitively difficult since it is not practical to  
781 place a large number (i.e., 100s) of accelerometers on this specific structure, let alone  
782 the effect of the added mass. By contrast, high-speed cameras provide a convenient way  
783 of measuring the structure without contact. Additionally, responses of the structure  
784 can be obtained with a high spatial resolution, making it possible to accomplish the  
785 identification of the whole structure.

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## 794 Data Access

795 The data used in the paper is available at [https://github.com/mattiacenedese/](https://github.com/mattiacenedese/BRBtesting)  
796 `BRBtesting`.

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